Towards the Next-Generation CFD:

High-order, fully *HP*-Adaptive Methods for 3-D Fluid Dynamics

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基本情况

研究领域:计算流体力学(理论,算法,实现),气体动力学,超级计算



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计算流体力学远景规划图(NASA CFD 2030)



长期致力于研究高精度 CFD 计算 所遇到的计算科学问题:

精度,效率,可用行,挑战性问题

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Towards **CFD 2030** with **HA3D**: a High-order HP-adaptive Discontinuous Galerkin code (by Shu-Jie Li, 2012-2019) (H= 网格, P= 精度, HP 自适应计算 +HP 多重网格)



HA3D: Parallel scalability

Strong scaling :



Weak scaling :

算例	计算规模	最小分	最大分	进程数	运行时	时间加	效率(加
		区(单元	区(单元		间(秒)	速比(基	速比/进
		数)	数)			于 24 核)	程数)
nacas2	79872*8	3062	3470	24	428s	1x	100%
nacas3	79872*64	2882	3460	24*8	439s	7.80x	97.5%
nacas4	79872*512	2186	3492	24*64	436s	62.83x	98.2%

Vectorization :



Memory usage :



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HA3D 的网格生成功能: Shu-Jie Li, *Mesh curving and refinement based on cubic Bézier surface for high-order discontinuous Galerkin methods*, Computational Mathematics and Mathematical Physics (2019 in press, a RAS (俄罗斯科学院) journal).

(部分发表在 Springer Book 《Numerical Geometry, Grid Generation and Scientific Computing》第 15 章,纪念 Voronoi 诞辰 150 周年,莫斯科, 2018 年 12 月)

论文中的9亿,混合四面体,六面体,金字塔的曲面网格生成



<u>算法到应用存在巨大 GAP:例如美国 2030 年计划实现</u> 发动机整机数值模拟(带燃烧)

Stanford University Center for Integrated Turbulence Simulations About CITS Research Projects Simulation Gallery Publications Software **Research Projects Research Projects** Simulation Framework | Software Engineering | Physics Modeling | Verification & Validation Integration Integrated Multicode Simulation Framework Merrimac CITS Home CHIMPS, Coupler for High-performance Integrated Multi-Physics Simulations CITS overarching problem is the simulation of the complete aero-thermodynamic flow path through a jet engine. CITS simulation environment is based on Python and built around the Center's flagship codes, CDP (Large Eddy Simulation approach) and TFLO (based on Reynoldsaveraged Navier-Stokes equations). It also includes a set of general-purpose interpolation and communication libraries (CHIMPS, Coupler for High-performance Integrated Multi-Physics Simulations) Basic concept: a parallel, scalable, distributed module that automatically takes care of mesh / solution interpolations to exchange information between CFD solvers, Structral dynamics codes, etc. The code-interfacing is designed around a distributed "interpolation server". Participating

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codes "register" the required data and their locations and "provide" their local mesh and solutions to the other components in the integrated environment. Information exchange (volume/overlap, surface/minimum distance and failsafe) happens transparently to the user through straightforward API.



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High order (spatial) methods are getting popular for the CFD community such as the Discontinuous Galerkin (DG), Flux Reconstruction (FR) ...

- 1. PROCS: spatially high accuracy
 - rich flow details
 - suitable for turbulence and aeroacoustic simulations (DNS, LES)
 - resolve boundary layer and vortex transportation
- 2. CONS: time marching could slow down the whole performance
 Total efficiency = Spatial discretization + TIME MARCHING
 计算速度,内存,鲁棒性,复杂度

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High-order solvers are expensive

Towards arbitrarily high-order, 3-D turbulence simulations **HA3D**: highly parallel, high-order Discontinuous Galerkin code (Author: Shu-Jie Li, 2012–present)

Strong Scaling on the TianHe2 supercomputer 100 23040 90 HA3D code (By Shuije Li CSRC) 80 19200 Mesh Partition Imbalance of the 2.3M mesh: Darallel Efficiency (%) 70 Min Partition: 125 Elements 15360 60 Speedup Max Partition: 159 Elements 50 11520 40 Ideal 7680 30 DG-P5 20 DG-P5 3840 DG-P6 10 DG-P6 4000 8000 10000 12000 14000 16000 2000 6000 Number of Processor



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Shortcomings of traditional time discretizations

HA3D has the following traditional options

- TVD-Runge Kutta, Jameson Runge Kutta (2-4th order)
- Point Jacobi
- Lower-Upper Symmetric Gauss Seidal (LU-SGS) and SGS
- Backword Differential Formula (BDF2)
- V-cycle *p*-multigrid (up to 11th order)
- Fully implicit restarted ILU-GMRES with exact Jacobians

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An inefficient time marching leads to massive time steps

- Severe time step restriction for viscous flows, e.g., Discontinuous Galerkin : $\Delta t_{viscous} = CFL \frac{\Delta x^2}{(p+1)^2 \lambda_{viscous}}$
- The term $(p+1)^2$ makes the time step even worse
- $\Delta t_{p=9}$ is at least 100X smaller than $\Delta t_{p=0}$ and suffering a dramatically drop of CFL number
- At least, we want to get rid of the term $(p+1)^2$ to make DG more efficient

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Outside Explicit and Implicit, the Exponential Schemes. Predictor-Corrector EXPonential time integrator (PCEXP)

- 任意时间步长,低绝对误差,时间准确,收敛快
 - S.J. Li, et.al, Adaptive Exponential Time Integration of the Navier-Stokes Equations.. (AIAA-2020 submitted)
 - S.J. Li, Efficient p-Multigrid Method based on Exponential Time Discretization for Compressible Steady Flows. arXiv:1807.01151
 - S.J. Li, et.al, An exponential time-integrator scheme for steady and unsteady inviscid flows. *Journal of Computational Physics* 365 (2018) 206–225
 - S.J. Li, et.al, *Exponential Time-Marching method for the Unsteady Navier-Stokes Equations.*. (*AIAA*-2019-0907)
 - S.J. Li, et.al, Fast Time Integration of Navier-Stokes Equations with an Exponential-Integrator Scheme. (AIAA-2018-0369)
 - S.J. Li, et.al, Explicit large time stepping with a second-order exponential time integrator scheme for unsteady and steady flows, (AIAA-2017-0753)

Basic ideal of Exponential schemes

$$\frac{\mathrm{d}\mathbf{u}}{\mathrm{d}t} = \mathbf{R}(\mathbf{u}) \tag{1}$$

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$$\frac{\mathrm{d}\mathbf{u}}{\mathrm{d}t} = \mathbf{J}_n \mathbf{u} + \underbrace{\mathbf{R}(\mathbf{u}) - \mathbf{J}_n \mathbf{u}}_{\mathbf{N}(\mathbf{u})}, \qquad \mathbf{J}_n = \frac{\partial \mathbf{R}(\mathbf{u})}{\partial \mathbf{u}} \qquad (2)$$

$$\mathbf{u}_{n+1} = e^{\Delta t \mathbf{J}_n} \mathbf{u}_n + e^{\Delta t \mathbf{J}_n} \int_0^{\Delta t} e^{-\tau \mathbf{J}_n} \underbrace{\mathbf{N}(\mathbf{u}(\mathbf{x}, t_n + \tau))}_{\text{numerical approximation}} d\tau$$
(3)

Basic ideal of Exponential schemes

$$\frac{\mathrm{d}\mathbf{u}}{\mathrm{d}t} = \mathbf{R}(\mathbf{u}) \qquad (4)$$

$$\frac{\mathrm{d}\mathbf{u}}{\mathrm{d}t} = \mathbf{J}_n \mathbf{u} + \underbrace{\mathbf{R}(\mathbf{u}) - \mathbf{J}_n \mathbf{u}}_{\mathbf{N}(\mathbf{u})}, \qquad \mathbf{J}_n = \frac{\partial \mathbf{R}(\mathbf{u})}{\partial \mathbf{u}} \qquad (5)$$

 $\mathbf{u}_{n+1} = e^{\Delta t \mathbf{J}_n} \mathbf{u}_n$, **EXACT** for linear equations (6)

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The PCEXP scheme S.-J. Li et al. AIAA-2017-0753, JCP-365(2018) 206-225

$$\mathbf{u}^{*} = \mathbf{u}_{n} + \Delta t \Phi_{1}(\Delta t \mathbf{J}_{n}) \mathbf{R}(\mathbf{u}_{n})$$
(7)

$$\mathbf{u}_{n+1} = \mathbf{u}^* + \frac{1}{2} \Delta t \, \Phi_1(\Delta t \mathbf{J}_n) \left[(\mathbf{N}(\mathbf{u}^*) - \mathbf{N}(\mathbf{u}_n)) \right] \quad (8)$$

where Φ_1 is a matrix function defined as

$$\Phi_{1}(\Delta t \mathbf{J}) = \frac{1}{\Delta t} \int_{0}^{\Delta t} e^{(\Delta t - \tau)\mathbf{J}} d\tau \qquad (9)$$
$$\stackrel{??}{=} \frac{\mathbf{J}^{-1}}{\Delta t} [\exp(\Delta t \mathbf{J}) - \mathbf{I}] \qquad (10)$$

Matrix Exponential : $\exp(\Delta t \mathbf{J}) := \sum_{m=0}^{\infty} \frac{(\Delta t \mathbf{J})^m}{m!}$ (11)

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How to compute Φ_1 times the vector such as **R**

$$\mathbf{u}^* = \mathbf{u}_n + \Delta t \, \Phi_1(\Delta t \mathbf{J}_n) \mathbf{R}(\mathbf{u}_n) \tag{12}$$

$$\mathbf{u}_{n+1} = \mathbf{u}^* + \frac{1}{2} \Delta t \, \Phi_1(\Delta t \mathbf{J}_n) \left[(\mathbf{N}(\mathbf{u}^*) - \mathbf{N}(\mathbf{u}_n) \right]$$
(13)

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Krylov subspace projection

$$\Phi_1(\Delta t \mathbf{J})\mathbf{R} = \mathbf{J}^{-1} \frac{e^{\Delta t \mathbf{J}} - \mathbf{I}}{\Delta t} \mathbf{R} = \left(\mathbf{I} + \frac{(\Delta t \mathbf{J})}{2!} + \frac{(\Delta t \mathbf{J})^2}{3!} + \frac{(\Delta t \mathbf{J})^3}{4!} + \cdots\right) \mathbf{R},$$
$$\mathcal{K}_m(\mathbf{J}, \mathbf{R}) = span\{\mathbf{R}, \mathbf{J}\mathbf{R}, \mathbf{J}^2\mathbf{R}, \cdots, \mathbf{J}^{m-1}\mathbf{R}\}.$$

PCEXP for unsteady flows: accuracy and cost



(a): Vortex transportation on a 24 × 24 uniform mesh with CFL = 0.1×2^n , $0 \le n \le 5$. Left: Temporal convergence of PCEXP, BDF2, and TVDRK3 schemes. Right: Evolution of the errors versus CPU time for PCEXP and BDF2 schemes. From top to bottom: $0 \le p \le 3$.

Comparison: TVDRK3 and BDF2 v.s. PCEXP



(A): Vortex transport on a 24×24 stretched mesh. Density Contours of 15 iso-lines in the range of [0.99691, 0.99974]. Top: BDF2 (black solid lines) *versus* TVDRK3 (red dashed lines). Bottom: PCEXP (blue solid lines) *versus* TVDRK3 (red dashed lines). From the left to the right: p = 1, 2, and 3.

表: Vortex transportation on 24×24 stretched mesh in one period. The t_P , t_B , and t_T are the CPU times (in minutes) corresponding to the PCEXP, BDF2, and TVDRK3 schemes, respectively.

PCEXP				BDF2			TVDRK3		
<i>p</i> order	Steps	CFL	t _P	Steps	CFL	$t_{\rm B}/t_{\rm P}$	Steps	CFL	$t_{\rm T}/t_{\rm P}$
p = 0	12	1000.0	0.03	12	1000.0	3.90	9696	1.2	48.97
p=1	35	1000.0	1.10	35	1000.0	1.13	29106	1.2	10.70
<i>p</i> = 2	52	1000.0	12.10	52	1000.0	1.00	48507	1.2	4.98
<i>p</i> = 3	84	1000.0	76.50	84	1000.0	1.60	67893	1.2	3.80

Unsteady viscous solution with PCEXP v.s. Analytical solution (rotating concentric cylinder flow)



B: Quasi-2D large-aspect-ratio mesh and velocity profiles comparsion (40 \times 40 \times 1 = 1600 cells)

Unsteady viscous solution with PCEXP v.s. Analytical solution (rotating concentric cylinder flow)

 $\overline{\ensuremath{\boldsymbol{k}}}$: Results statistics of the PCEXP scheme for the rotating concentric cylinder flow.

Time ($T = r_0^2 / \nu$)	0.2 T	1.0 <i>T</i>	2.0 T	4.0 <i>T</i>
#Time steps	16	79	158	315
CFL	10 ³	10 ³	10 ³	10 ³
L_2 Error (%)	0.28	0.24	0.26	0.29

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Unsteady laminar flows past a circular cylinder



(a) Re = 60



(b) Re = 180

图: Vorticity field for the laminar flow past a circular cylinder with CFL = 500.0 a @

表: Results statistics for the laminar flow past a circular cylinder.

Re	60	80	100	120	140	160	180
Present	0.139	0.155	0.168	0.178	0.185	0.191	0.197
Experiments	0.135	0.152	0.164	0.173	0.181	0.186	N/A
Karniadakis	0.153	0.168	0.178	0.185	0.192	0.197	0.203

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Flow past a sphere at Ma = 0.3, Re = 300



(a) Q contour colored by the velocity magnitude



(b) Velocity magnitude contour on the cut slice at z = 0

Method	#Cell	S _{order}	T _{order}	C _d	ΔC_d	St
Present	80093	3rd	2nd	0.674	0.0033	0.133
Gassner	160000	4th	4th	0.673	0.0031	0.131
Johnson	428442	2nd	2nd	0.656	0.0035	0.137
Haga	54312	4th	3rd	0.670	0.0032	0.131

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 $\overline{\mathbf{x}}:$ Results statistics of the PCEXP scheme for the flow past a sphere at $\mathrm{Re}=300.$

Turbulent flow past a square cylinder at Ma = 0.15, Re = 22000



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图: Turbulent flow field for the flow past a square cylinder: Ma = 0.15, Re = 22000, DG p = 4.

多体运动计算:动态自动网格剖分(h remeshing)

三维网格**实时劈分,边交换,拓扑优化,** 自适应全结合 _{初步实现:弹体发射}

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p 高阶自适应算法应用在螺旋桨:自动捕捉需要高精度 计算的区域(AIAA paper 2020, submitted)



单级叶片,周期边界的高阶快速计算 (by Shu-Jie Li, 08-16-2019) (指数时间 +_间断有限元)



中物院与中国航空发动机集团全面战略合作子方向

复杂几何边界上的湍流大涡模拟



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Summary

- 在世界范围内首次实现了 hp 自适应指数时间推进高精度求解三维
 流动问题
- ❷ 支持旋转,多体运动,网格,精度(hp)等全套自适应及多级加速 算法,MPI/OPENMP/OPENACC 混合并行
- ②发展了一整套三维任意精度阶,工业级流动模拟方法(包含曲面网格生成,大规模计算,远程IO并行可视化)

Ongoings

❶ 发展旋转流动的高精度先进计算方法

- S.-J. Li. arXiv:1807.01151 (p-multigrid exponential for steady)
- S.-J. Li et al. AIAA-2019-0907 (3-D unsteady NS)
- S.-J. Li et al. AIAA-2018-0369 (3-D steady NS)
- S.-J .Li et al. JCP, 365(2018) 206-225 (3-D steady, unsteady inviscid)
- S.-J. Li et al. AIAA-2017-0753 (3-D steady, unsteady inviscid)

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